## **Torque & Fastener Matching Information**

In most race car chassis and aircraft airframe applications, the bolts used are loaded in shear. The torque values given in the NAS torque chart (opposite page) are adequate and appropriate for this type of usage. For aerospace bolt applications subjected to high tensile loadings and for aerospace nuts and bolts matched to fasteners of other grades and thread classes, see the notes below:

- A.) For applications requiring high installation torque values and using standard NAS bolts: achievable torque is dependent on plating, length of thread engagement, class of mating threads and lubricant (if any). For maximum fatigue resistance and clamping force in the bolted assembly, we want to produce a level of installed tensile stress that is just below the yield strength of the bolt material (typically around 85% of the ultimate tensile strength). Achieving this ensures that the stresses encountered in the assembly/tightening process will be the most the bolt ever sees. At the yield point, the bolt will undergo "plastic" deformation (permanent stretch). Bolts for critical applications should be test-torqued to determine the point of measurable permanent stretch. As a rule, approximately 90% of this value will give the best performance in a bolted assembly. Example: '93 INDY-LOLA ring gear bolts NAS 1307-4H with light lubrication, these bolts stretch permanently on a consistent basis between 105 and 110 ft/lbs of installed torque. 90% of this value gives an assembly torque of 95 ft/lbs.
- B.) The mating of aerospace spec (MS•NAS•AN) bolts and nuts to fasteners of other specifications and thread classes can cause problems. Nut splitting failures are primarily caused by overtorquing and/or thread class mismatches. Generally, NAS wrenching torque values should be reduced when aerospace nuts are matched with non-aerospace bolts or studs. Wrench torque values in the NAS nut tables below are achieved under laboratory conditions using aerospace spec bolts (Class 3 threads) of higher strength than the nuts they are mated to. THE TABLES DO NOT STATE, NOR SHOULD ONE INFER THAT THIS PERFORMANCE CAN BE DUPLICATED WITH OTHER TYPES OF FASTENERS AND/OR CLASSES OF THREADS. For applications requiring high clamping forces, 12 point nuts should be used whenever possible. Their greater thread contact area and higher rated strength make higher wrenching torque values possible. As a general rule, the wrenching torque values for aerospace spec nuts should be reduced approximately 25% as a starting point when these nuts are mated to non-aerospace bolts or studs.

NUT SIZE	WRENCH TORQUE, INCH POUNDS 160 KSI NAS NUTS *	WRENCH TORQUE, INCH POUNDS 180 & 220 KSI NAS NUTS *							
#6 - 32	20	N/A							
#8 - 32	30	N/A							
#10 - 32	60	70 / 85							
1/4 - 28	150	170 / 210							
5/16 - 24	330	370 / 450							
3/8 - 24	530	600 / 730							
7/16 - 20	825	880 / 1100							
1/2 - 20	1125	1225 / 1400							
9/16 - 18	1550	1700 / 2000							
5/8 - 18	2000	2200 / 2600							
3/4 - 16	N/A	3800 / 4400							
* Above values for alloy steel nuts. For A286 stainless multiply them by 0.694									

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## Recommended Torque Values for Nut & Bolt Combinations 31

for shear loads without lubrication

		Steel Ten	sion Rolts			Steel Ten	sion Rolts			Aluminu	n Rolts	
	Steel Tension Bolts  AN3 through AN20 AN42 through AN49 AN73 through AN81 MS20033 through MS20046 MS20073 and MS20074 AN509 AN525 MS24694 and MS27039 Steel Nuts				Steel Tension Bolts  MS20004 through MS20024  NAS144 through NAS158  NAS333 through NAS340  NAS583 through NAS590  NAS624 through NAS644  NAS1303 through NAS1320  NAS172 and NAS174  NAS517			Aluminum Bolts  AN3DD through AN20DD AN173DD through AN186DD AN509DD AN525DD MS27039D MS24694DD  Aluminum Nuts				
	Tension			Shear			el Nuts Shear		Tension		Shear	
	AN310 & AN315 AN363 & AN365 NAS1021 MS20365 MS21042 MS21045 NAS679		AN: AN: NAS MS1 MS2	320 364 1022 7826 0364	Tension  AN310 & AN315 AN363 & AN365 NAS1021 MS20365 MS21042 MS21045 NAS679		AN320 AN364 NAS1022 MS17826 MS20364		AN310D AN365D NAS1021D		AN320D AN364D NAS1022D	
	_	_			or Fine Th					_		_
Series	From	To	From	То	From	То	From	То	From	To	From	To
#8-36 #10-32	12 20	15 25	7 12	9 15	25	30	15	20	5 10	10 15	3 5	6 10
1/4-28 5/16-24	50 100	70 140	30 60	40 85	80 120	100 145	50 70	60 90	30 40	45 65	15 25	30 40
3/8-24 7/16-20	160 450	190 500	95 270	110 300	200 520	250 630	120 300	150 400	75 180	110 280	45 110	70 170
1/2-20 9/16-18	480 800	690 1000	290 480	410 600	770 1100	950 1300	450 650	550 800	280 380	410 580	160 230	260 360
5/8-18 3/4-16	1100 2300	1300 2500	660 1300	780 1500	1250 2650	1550 3200	750 1600	950 1900	550 950	670 1250	270 560	420 880
7/8-14 1-14	2500 3700	3000 4500	1500 2200	1800 3300	3550 4500	4350 5500	2100 2700	2600 3300	1250 1600	1900 2400	750 950	1200 1500
1 1/8-12 1 1/4-12	5000 9000	7000 11000	3000 5400	4200 6600	6000 11000	7300 13400	3600 6600	4400 8000	2100 3900	3200 5600	1250 2300	2000 3650

## **Customized & Modified Aircraft & Aerospace Bolts**

Shortening, drilling, grinding, machining, thread rolling to your print or specifications. From standard AN•MS•NAS fasteners. Contact us for technical assistance and recommendations. Combinations of the above operations can be simple or elaborate. For example, a 7/16 diameter MS21250 bolt (pp. 22-23) was used to manufacture custom, rolledthread (J-type), metric (M10 x 1.0) transmission bolts with heads drilled for safety wire and head base diameter turned down for clearance. Total turn around time was four weeks for a fastener of higher quality than the German-made equivalent at one-third to one-half the price.

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